



US Department of Transportation  
Federal Aviation Administration

**MAJOR REPAIR AND ALTERATION**  
**(Airframe, Powerplant, Propeller, or Appliance)**

Form Approved  
OMB No. 2120-0020  
**For FAA Use Only**  
Office Identification

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

1. Aircraft	Make Stinson	Model HW75 (105)
	Serial No. 7247	Nationality and Registration Mark N23792
2. Owner	Name (As shown on registration certificate) Twenty-One-O- Three Mike Corp	Address (As shown on registration certificate) C/O Dale P Jewett 2601 Canterbury Drive Hutchinson, Kansas 67502

**3. For FAA Use Only**

**4. Unit Identification**

**5. Type**

Unit	Make	Model	Serial No.	Repair	Alteration
AIRFRAME	(As described in Item 1 above)				X
POWERPLANT					
PROPELLER					
APPLIANCE	Type				
	Manufacturer				

**6. Conformity Statement**

A. Agency's Name and Address Alvin L Sowers 1307 E 26th Hutchinson, Kanasa 67502	B. Kind of Agency <input checked="" type="checkbox"/> U.S. Certificated Mechanic <input type="checkbox"/> Foreign Certificated Mechanic <input type="checkbox"/> Certificated Repair Station <input type="checkbox"/> Manufacturer	C. Certificate No. A & P 1514515
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D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date 5-6-03	Signature of Authorized Individual <i>Alvin L Sowers</i>
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**7. Approval for Return To Service**

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is  APPROVED  REJECTED

BY	FAA Fit. Standards Inspector	Manufacturer	<input checked="" type="checkbox"/>	Inspection Authorization	Other (Specify)
	FAA Designee	Repair Station	<input checked="" type="checkbox"/>	Person Approved by Transport Canada Airworthiness Group	
Date of Approval or Rejection 5-8-03		Certificate or Designation No. 1514515	Signature of Authorized Individual <i>Alvin L. Sowers</i>		

## NOTICE

*Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.*

### **8. Description of Work Accomplished**

*(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)*

Removed all old fabric coverings and installed all new fabric as listed below. Inspections & repairs to all covered units are listed in the airframe log book # 3.

Fuselage, wings, ailerons, wing flaps, elevator, rudder and horizontal stabilizer were recovered with the Poly-Fiber Inc system using Poly-Fiber covering kit No (1) in accordance with STC SA1008WE.

Recovering completed per the procedure manual # 1 Revision No. 19 dated July 1999.

Checked static balance on both aileron control surfaces.

Continued airworthiness is included in the Poly-Fiber procedure manual # 1 Revision 19 Appendix A through H which is included with the above listed procedures manual

Additional Sheets Are Attached